

specific heat ratio $\gamma = 1.32$.

specific heat at constant pressure $C_p = 80 \text{ J/molK}$.

for isentropic expansion 3-4 relation between temperature and pressure can be given as :

Since the flow is also isentropic,

$$\frac{p_2}{p_1} = \left(\frac{T_2}{T_1} \right)^{\frac{\gamma-1}{\gamma}}$$

$T_3 = 333 \text{ K}$, $P_3 = 25 \text{ Bar}$, $P_4 = 4 \text{ Bar}$.

solving gives, $T_4 = 213.56 \text{ K}$ (outlet temperature).

power output $= 6 \times 10^9 \text{ watt}$

net power output = turbine power - pump power.

power difference can be written in the form of enthalpy change. $m = \text{mass flowrate}$.

$= m \cdot C_p \cdot (T_3 - T_4 - T_2 + T_1)$.

$T_4 = T_1$ (isothermal heat rejection)

$6 \times 10^9 = m \cdot C_p \cdot (T_3 - T_2)$

Since no further information is given assuming pump work to be negligible.

$6 \times 10^9 = m \cdot C_p \cdot (T_3 - T_4)$

$6 \times 10^9 = m \cdot 80 \cdot (333 - 213.56)$

$\Rightarrow m = 627930 \text{ kg}$.

This has been cut and pasted, the last part should be in moles not Kg. as ammonia has molar mass of 17 grams per mole, the mass is grams $= 106 \times 10^5 \text{ gram}$
 $= 106 \times 10^2 \text{ Kg} = 10.6 \text{ Tonnes per second}$.

Would this answer, 10.6 tonnes per second be correct?

The exit temperature of -60 at 4 atmospheres will be a liquid...can a turbine work when it's exhaust gas is a liquid? I am thinking turbine exhaust points downward into a vacuum chamber which collects liquid at base.